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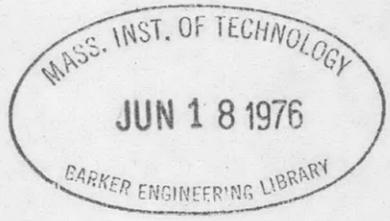
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NAVY DEPARTMENT  
DAVID TAYLOR MODEL BASIN  
WASHINGTON, D.C.

CALCULATION OF FLEXURAL CRITICAL FREQUENCIES OF SHIP HULLS  
BY PROHL'S METHOD

by



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## ABSTRACT

This report extends Prohl's method for the calculation of critical speeds of flexible rotors to calculating flexural critical frequencies of ships. The method is simplified and set up so that the computations can be readily made with a punched card machine. The method is applied to a uniform bar as a check.

## AUTHORIZATION

This study was undertaken under the general project established by Bureau of Ships letter S87-(8)(330) of 4 August 1943 for the development of techniques and instruments for observation of motions on ships.

## INTRODUCTION

Most of the methods for obtaining the critical frequencies of ship hulls give only the frequency in the lowest mode of vibration. When attempts are made to apply them to higher modes either the accuracy is greatly reduced or the methods themselves are not applicable.

An accurate method for the calculation of critical frequencies in the higher modes would be very useful, provided the value of the method were not outweighed by the extent of the computations involved. However, it must be kept in mind that all such calculations when applied to ships involve in addition to the mass of the ship itself a quantity representing the virtual mass of the surrounding water. Although theories exist for the estimation of the virtual mass, References (1) and (2), experimental verification of the theories is still unsatisfactory. The method given in this report is based on a given mass-distribution curve for the ship which in this case must include the virtual mass of the water. The extent to which any method of calculating flexural frequencies of ships is limited by the uncertainty as to the virtual mass of the water must remain for the present a matter of individual judgment.

## THEORY OF METHOD

The numerical calculation of critical speeds of flexible rotors as given by M. A. Prohl, Reference (3), may be ex-

tended to the calculation of flexural critical frequencies of ships. If the ship is assumed to be vibrating freely in one of its normal modes and at a circular frequency  $\omega$ , the deflection at the instant of maximum displacement from the rest position considered as a function of distance from one end must satisfy the differential equation:

$$\frac{d^2}{dx^2} \left( EI \frac{d^2 y}{dx^2} \right) = m \omega^2 y$$

where

- E is the modulus of elasticity, tons/ft.<sup>2</sup>,
- I is the variable moment of inertia of cross-section of ship, ft.<sup>4</sup>,
- x is the distance along the length of ship, ft.,
- y is the deflection normal to longitudinal axis of ship, ft.,
- m is the variable mass per unit length of ship at distance x from end, tons sec.<sup>2</sup>/ft<sup>2</sup>,
- $\omega$  is the circular critical frequency of the vibration, radians/sec.  
( $2\pi$  times the frequency of flexural vibration in cycles per second)

There are four boundary conditions which must be satisfied for this fourth-order differential equation, and any frequency  $\omega$  which will give a solution of the equation satisfying the four boundary conditions is a critical frequency.

In Reference (3), Prohl treats the case of a turbine-rotor shaft supported in two bearings and overhanging at one end. This imposes the condition of zero deflection at either of the bearings. Furthermore, by the assumption of simple support at the bearings Prohl obtains the condition of zero moment at the end that does not overhang. In the case of a ship the points of zero deflection are the nodal points whose location cannot be assumed to be known. The four known end conditions are that the moment and shear must be zero at each end, as no external forces or couples are assumed acting. If the ship is assumed to be vibrating with unit amplitude at one end and at a circular frequency  $\omega$ , the shear and moment being taken as zero at that end, then in order for  $\omega$  to represent a critical frequency it must have such a value that the shear and moment also become zero at the other end.

Prohl simulates the actual rotor by a lumped rotor in which the mass is concentrated at a large number of equally spaced points whereas the bending stiffness is concentrated in massless members connecting the lumped masses. This enables him to proceed step by step from one end of the shaft to the other.

In order to apply Prohl's method to a ship, the latter can be divided into twenty or more sections of equal length and the weight of each section can be lumped at the ends of the section, half at each end. The section ends are designated as stations and hence there are twenty-one stations for the twenty sections. Figure 1 shows the hull of a ship in its fundamental mode of vibration with the masses concentrated at twenty-one stations. Unit deflection is assumed at the left end of the ship, that is, Station 0, with other boundary conditions at this end following the laws of dynamics for a free-free bar, that is the bending moment and shearing force must be zero. As the slope at the end cannot be assumed known, its value, designated as  $\theta_0$ , must be carried through the calculation as an unknown quantity. With the necessary boundary conditions known or assumed, the calculation proceeds for some chosen value of  $\omega$ .

The following nomenclature is used in the calculation

V is shearing force in tons  
M is bending moment in ton-ft.  
y is deflection normal to longitudinal axis in feet  
 $\Delta x$  is length of each section in feet

The computations involve also increments of these quantities.

By virtue of the fact that the ship is structurally continuous, the deflection curve for the ship will also be continuous; thus the value of deflection will be the same for the right end of Section 1 and the left end of Section 2, etc. For each station there is a process of four integrations in deducing the deflection at the station from the deflection at the preceding station; at each section are added increments of shear, bending moment, slope, and deflection until the calculation is completed for the length of the ship. The increments are designated by  $\Delta$ 's.

The following formulas as well as the tables to follow will help clarify the above discussion:

$$\Delta V_n = m_{n-1} y_{n-1} \omega^2$$

$$V_n = V_{n-1} + \Delta V_n$$

$$\Delta M_n = V_n \Delta x$$

$$M_n = M_{n-1} + \Delta M_n$$

$$\Delta \theta_n = \frac{\Delta x}{EI} M_n$$

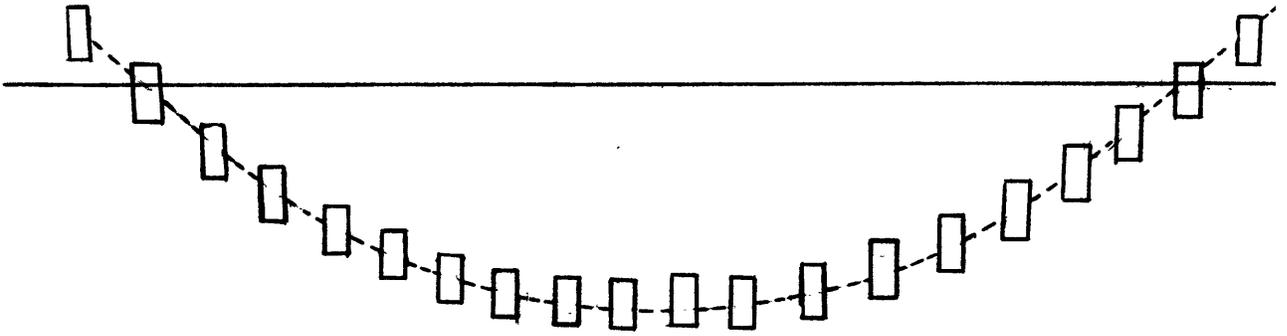
$$\theta_n = \theta_{n-1} + \Delta \theta_n$$

$$\Delta y_n = \theta_n \Delta x$$

$$y_n = y_{n-1} + \Delta y_{n-1}$$

where n designates the number of the station from 0 to 20A.

Since each term is of the form  $a + b\theta_0$ , where a and b are real numbers and since the assumed conditions at the left end are that  $y = 1$  and  $\theta = \theta_0$  (an unknown value),  $y_1$ , expressed in the form  $a + b\theta_0$ , has the value  $1 + 0 \cdot \theta_0$ . A sample table for starting the calculations on a uniform bar is given on page 5. Since all computed quantities in this table will be of the form  $a + b\theta_0$  and since the operations required to obtain one term from a preceding one are merely multiplications and additions, the "a" and "b" terms may be treated independently. Hence, it is simpler to make the computation by means of two separate tables, one for the "a" terms and one for the "b" terms. Thus, for Station 0 in the "a" table the value of y is 1.0 and the value of  $\theta$  is 0 whereas in the "b" table the value of y is 0 and the value of  $\theta$  is 1.0. The true value applicable to any of the spaces in the table on page 5 is



*Figure 1 - Fundamental Mode of Vibration of the Hull of a Ship with Masses Concentrated at Twenty One Stations*

Table 1 - Sample Table Showing the Initial Data for Calculating the Resonant Frequencies of a Uniform Bar

Station	m	y	$\Delta V$	V	$\Delta M$	M	$\frac{\Delta x}{EI} \times 10^{12}$	$\Delta \theta$	$\theta$	$\Delta$
0	38.8	$1+0.0_0$		$0+0.0_0$		$0+0.0_0$			$0+1.0_0$	$0+$
1	77.5						899			
2	77.5						899			
3	77.5						899			
4	77.5						899			
5	77.5						899			
6	77.5						899			
7	77.5						899			
8	77.5						899			
9	77.5						899			
10	77.5						899			
11	77.5						899			
12	77.5						899			
13	77.5						899			
14	77.5						899			
15	77.5						899			
16	77.5						899			
17	77.5						899			
18	77.5						899			
19	77.5						899			
20	38.8						899			
20A										

obtained by adding the corresponding values in the "a" and "b" tables and expressing the result in the form  $a + b\theta_0$ . Although dividing the ship into twenty segments with the station at the left end designated as 0 makes the station at the right end Station 20, the lumping of masses at the stations makes it necessary to consider an auxiliary Station 20A also. The values of shear and bending moment listed in the table on page 5 are the values which the section to the left exerts on the section to the right of the station in accordance with the ordinary conventions adopted in simple-beam theory. The shear load at Station 20A differs from that at 20 due to the inertia effect of the mass assumed concentrated at 20, that is,  $V_{20A} = V_{20} + y_{20} m_{20} \omega^2$ . The moment however, does not change abruptly, being a continuous quantity.

When the values for shear and bending moment for Station 20A have been obtained in the form  $a + b\theta_0$ , it is necessary to determine some relationship between the two values before the critical frequencies can be estimated. The residual shear and moment at the forward end of the ship are written in the form

$$V_{20A} = c + d\theta_0$$

$$M_{20A} = e + f\theta_0$$

where for Station 20A, c is the value of shear computed in the "a" table and d the value in the "b" table. Since the moment does not change at a point of concentrated mass  $M_{20A}$  and  $M_{20}$  are identical.

If the assumed frequency is a critical frequency then

$$V_{20A} = c + d\theta_0 = 0$$

and

$$M_{20A} = e + f\theta_0 = 0$$

Therefore  $\theta_0 = -c/d = -e/f$  and the natural frequencies occur when  $c/d = e/f$ . Hence, if  $c/d$  and  $e/f$  are plotted against the assumed frequencies, the critical frequencies will occur at the points of intersection of the curves.

It is simpler, however, to eliminate the plotting of the two curves by assuming either the shear or bending moment to

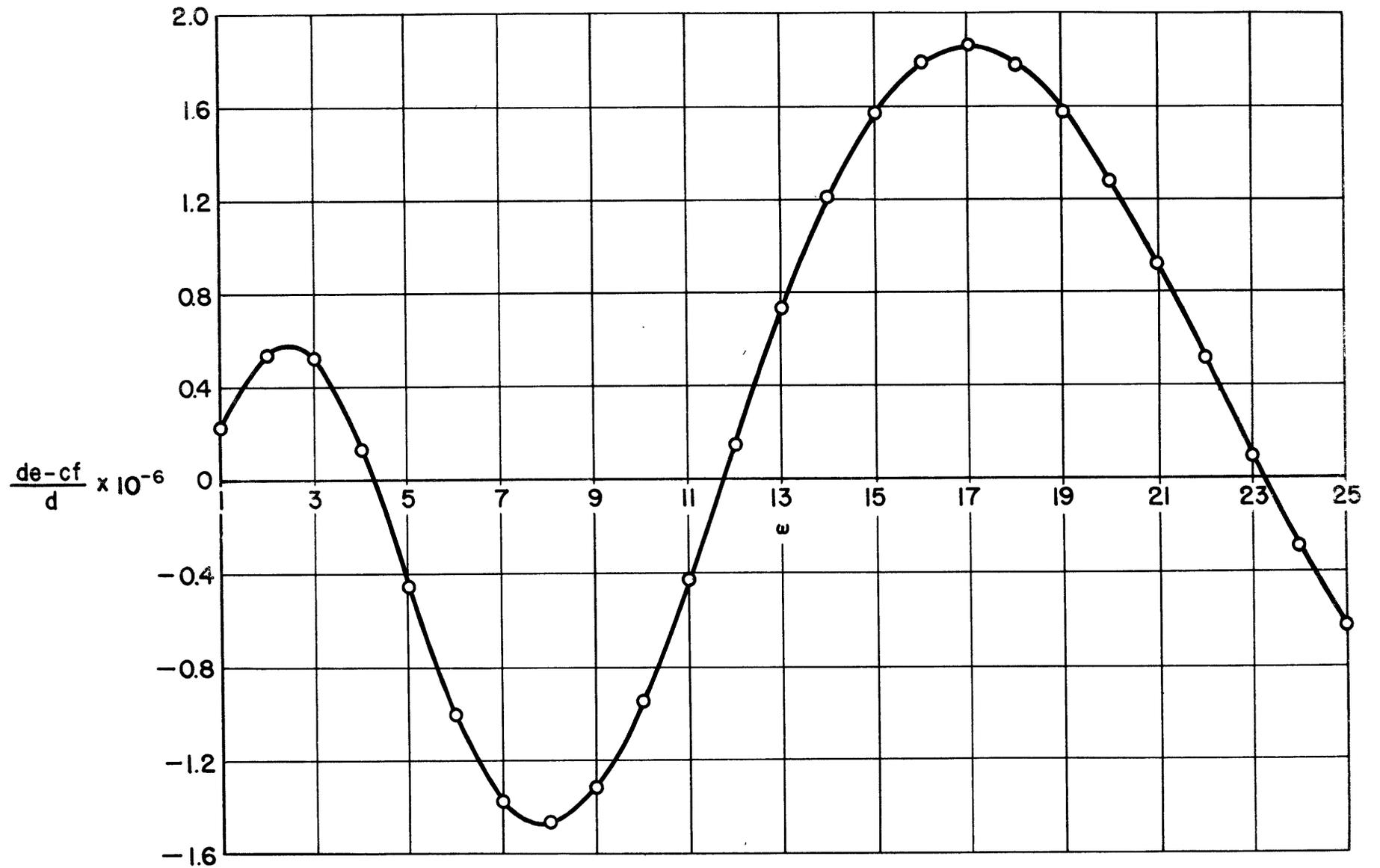


Figure 2 - Calculation of Natural Flexural Frequencies of Uniform Bar by Prohl's Method  
 (The criticals occur at the values of  $\omega$  where  $\frac{de-cf}{d}$  is equal to zero)

be zero:

$$V_{20A} = c + d\theta_0 = 0$$

$$M_{20A} = e + f\theta_0$$

Then  $M_{20}$  can be written in the form

$$M_{20A} = \frac{de - cf}{d},$$

which shows that  $M_{20A}$  is equal to zero only when  $c/d = e/f$ .

Therefore  $\frac{de - cf}{d}$  is plotted against the assumed frequencies and where the curve crosses the zero line is a critical frequency.

#### APPLICATION OF METHOD TO A UNIFORM BAR

In order to check the accuracy of the method, calculations were made for a uniform bar since in this case exact formulas are available for the natural frequencies. The initial data for making the calculation for the uniform bar are given in the table on page 5 where it will be observed that the "a" and "b" terms are included in the same table even though they can be treated separately. A typical punched-card machine calculation made from initial data for one of the assumed values of  $\omega$  is given on page 8.

Similar calculations were made for frequencies from  $\omega = 1$  through  $\omega = 25$ . The value for

$$M_{20A} = \frac{de - cf}{d}$$

was calculated for each  $\omega$  assumed and the data were plotted in Figure 2. The critical values obtained from this curve are  $\omega = 4.25$ ,  $\omega = 11.74$ ,  $\omega = 23.25$ . These values check with the critical frequencies as computed by the well-known formulas for the uniform bar.

A punched-card type of calculating machine is believed to be the only practicable means of handling the calculations involved in this method. With the use of such a machine the time necessary for the calculations is reduced from several months to a few days. It appears from the results with the uniform bar calculation that the method is

initial values					station			Y		V		M		θ	
ω	Y	V	M	θ		m	ΔX								
16	0	0	0	1	0	388	50							1000000000	0000000000
16					1	775	50	50	00000000					1000000000	0000000000
16					2	775	50	1022	295200	99200000		49600000		1044890400	0000000000
16					3	775	50	1634	4765352	302023368		200611684		1224940304	0000000000
16					4	775	50	2478	185049	626360814		513792091		1686839394	0000000000
16					5	775	50	3803	3832147	1118038728		1072808458		2681224122	0000000000
16					6	775	50	6032	2598898	1872713026		2009164968		4487833301	0000000000
16					7	775	50	9854	3738553	3069580647		3543958292		7643549309	0000000000
16					8	775	50	1639	8454820	5024688360		6056299472		13088162834	0000000000
16					9	775	50	2752	3355069	8278141796		10195370370		22853800497	0000000000
16					10	775	50	4632	2953260	13739172242		17064986491		37598196388	0000000000
16					11	775	50	7794	4687371	22929646169		28529779572		63243462221	0000000000
16					12	775	50	1310	19580166	38393872143		47726718648		106149785889	0000000000
16					13	775	50	2200	18869896	64388186848		79920794072		177998579460	0000000000
16					14	775	50	3698	24884830	108039900635		133940744390		298411300669	0000000000
16					15	775	50	6193	2380060	181294046242		224587767511		500318711659	0000000000
16					16	775	50	10388	56782359	304179810446		376677522734		838948204899	0000000000
16					17	775	50	17423	35115568	510288696066		631821870767		140695662221	0000000000
16					18	775	50	2922	2196183865	855967982995		1059805862265		2359722136893	0000000000
16					19	775	50	4901	20688148	1435731705874		1777671715202		3957849008560	0000000000
16					20	775	50	8220	332261236	2408114050402		2981728740403		6638423146288	0000000000
16					20a	388	50	8220	332261236	3224623213246					
(d)															
16	1	0	0	0	0	388	50	1000	000000					0000000000	0000000000
16					1	775	50	1022	23240	993280		496640		000446479	0000000000
16					2	775	50	1134	8818	3021371		2007428		002281188	0000000000
16					3	775	50	1456	1880	5273176		4644014		006426224	0000000000
16					4	775	50	2169	6893	8162253		8725141		014270026	0000000000
16					5	775	50	3555	3797	12466917		14958600		027717807	0000000000
16					6	775	50	6052	5978	19521187		24719194		049940362	0000000000
16					7	775	50	1036	93702	31529541		40483965		086338447	0000000000
16					8	775	50	1767	68976	52102371		66535151		146150848	0000000000
16					9	775	50	2993	44008	87173336		110121619		245150063	0000000000
16					10	775	50	5043	38874	146863187		183403413		410029731	0000000000
16					11	775	50	8472	43214	246627988		306717407		685768680	0000000000
16					12	775	50	1421	205583	414721042		514077928		1147224737	0000000000
16					13	775	50	2382	286660	696688230		862422043		1923242154	0000000000
16					14	775	50	3994	938319	1169441039		1447142563		3224223318	0000000000
16					15	775	50	6698	508332	1962036801		2428160964		5407140028	0000000000
16					16	775	50	1123	3193635	3291020854		4073671391		9069370606	0000000000
16					17	775	50	1883	9543763	5519686471		6833514687		15212700256	0000000000
16					18	775	50	3159	8171043	9257451954		11462240604		22517224252	0000000000
16					19	775	50	5299	8662887	15526829089		19225805149		42600983688	0000000000
16					20	775	50	8889	3838286	26041463806		32246237052		71790380798	0000000000
16					20a	388	50	8889	3838286	34871110975					
(e)															

Table 2-Sample Punch Card Machine Calculation for a Uniform Bar for an Assumed Frequency  $\omega = 16$  (vertical lines denote decimal points)





3 9080 02754 0696

reliable for flexural modes of free-free bars at least up to the 4-noded mode. The value of the method for use in ship design may be expected to increase as more experimental data are obtained as to the virtual mass of the surrounding water and as to the extent to which flexural vibration theory can be applied to ship hulls considered as built-up girders of varying section.

#### REFERENCES

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- (3) "A General Method for Calculating Critical Speeds of Flexible Rotors," by M. A. Prohl, Journal of Applied Mechanics, vol. 12, No. 3, September 1945, pp. A-142 - A-148.

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